

- 1. FLIGHT NO 47, ARTICLE 1801, FLOWN THURS 17 DEC 1964.
- 25X1A
 - 3. DURATION: 8:55 HRS, TOTAL TIME 56:50 HRS.
 - 4. MAX SPEED REACHED: 3:15 M, MAX ALT 73,000 FT
 - 5. TIME FROM 0.9 M TO 1.2 M: NOT AVAILABLE. WAS 3.15 ON FLIGHT 46, 16 NOVEMBER.
 - 6. TIME FROM 1.2 M TO 2.8 M: NOT AVAILABLE.
 - 7. TIME ABOVE 2.0 M: 0/25 HRS THIS FLIGHT. TOTAL 5:51 HRS.
 - 8. TIME ABOVE 2.6 M: 8:15 HRS THIS FLIGHT. TOTAL 8:57 HRS.
 - 9. TIME ABOVE 3.0 M: 2:12 PRS THIS FLIGHT. TOTAL 0:30 HRS.
 - ID. TIME ABOVE 3.0 M: 0:08 HRS THIS FLIGHT, TOTAL 0:13 HRS.
 - 11. T.O. WEIGHT: 118,871 LBS., C.G. 21.54 PERCENT.
 - 12. CONFIGURATION: SAME AS FOR FLIGHT 46.
 - 13. OBJECTIVES: ACCELERATION TO MAX MACH. EVALUATION OF
 - 425 KEAS CLIMB SCHEDULE. INS EVALUATION.

REPRODUCTION BY DTHER

14. CLIMB WAS MADE TO THE EAST FROM EAFB, WITH TURN AROUND

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NEAR PHOENIX ARIZONA.

- 15. CLIMB SCHEDULE WAS AS FOLLOWS: MAF A/B TAKEOFF; 50

 PERCENT A/B UPON REACHING 400 KEAS; THIS CONDITION HELD TO 0.9 M;

 AT 0.9 POWER TO FULL A/B FOR PEST OF ACCEL, SPEED INCREASED TO

 425 KEAS; SPEED REDUCED TO 400 KEAS AT 2.0 M FOR BALANCE OF ACCEL.
- 16. 35 DEGREE BANK TURN STARTED AT 2.2 M, NOPTH OF PHOENIX. SPEED INCREASED IN TURN TO APPROX 3.8 M AT ROLL OUT ON RETURN COURSE.
- 17. SOME SLIGHT YAW OSCILLATION OBSERVED AT APPROX 2.95 M IN TURN. THIS WAS NOT OBSERVED AFTER POLL OUT ABOVE 3.6 M.
- 18. ACCELERATION WAS CONTINUED TO 3.15 M, WITH SMOOTH ENGINE AND INLET OPERATION ALL THE WAY. PILOT DISCONTINUED AT THIS POINT DUE TO CIT READING LIMIT OF 425 DEGPEES CENTIGPADE.
 REPORTED SOMETHING OVER 20,000 LBS FUEL REMAINING AT MAX CONDITION.
- 19. PH CIT SYSTEM MALFUNCTIONED DUPING DECELEPATION, APPARENTLY AS RESULT OF BPOKEN WIRE.
- 20. INS SET UP UNEVENTFUL. NO PROBLEM EXPERIENCED ON TRANSFER TO VEHICLE POWER. (BROKEN WIPE IN BAY FOUND TO HAVE CAUSED PROBLEM ON PREVIOUS FLIGHT). SYSTEM OPERATION VERY GOOD THROUGHOUT THE FLIGHT. TERMINAL EPROP APPROX 3NM AFTER 73 MINUTES IN NAV MODE.
- 21. DMAG CHUTE BROKE AWAY UPON PEACHING FULL DEPLOYMENT
 AFTEP LANDING. INVESTIGATION SHOWED THAT SMALL LATCH ON DOOP
 WAS FREE TO FLOAT UP APPROX 3/8 INCH IN FLIGHT. THIS ACTED
 AS SMALL AIP SCOOP, DIPECTING HEATED AIP ONTO SHOUD LINES
 AND BALL FITTING. SEVERAL OF TOPMOST SHROUD LINES OVERHEATED,
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PESULTING IN THEIR FAILING UNDER LOAD, WITH REMAINDER FAILING WHEN LOAD SHIFTED TO THEM. STEPS BEING TAKEN TO PIG THE LWTCH MORE SECURELY, AND PROVIDE MOPE EFFECTIVE SEALING AGAINST HOT AIR ENTRY INTO THE CHUTE COMPARTMENT.

END OF MESSAGE